June-07-12 9:22							3		
Item ID: Revision ID:	D412-698-041		Accept	*N900040	100	* Seti	ıp Start	*NS	31*
Item Name:	Hinge Panel Automatic Door Open	er, LH (Standard)					Stop	*N!	32*
Start Date: Required Date:	07/06/2012 Start Qty: 1.00 21/06/2012 Req'd Qty: 1.00	•		Cust Item ID: Customer:				1 41	12
Reference:									
Approvals:	Process Plan: _ MLJ QC:	Date: 12 06 C	Tooling: SPC (Y/N):	Date:		Rur	Start Stop	*NF	R1*
Sequence ID/ Work Center II	Operation		Set Up/ Run Hours	Tool ID Tool #		_		Reject Number	Insp. Stamp
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Dart Aerospace Ltd

W/O:			WO	RK ORDER CHANGI	ES				
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Part No	·	PAR #:	Fault Categ	jory:	NCR: Yes	No DQ	A:	Date: _	
- 44	Re	esolution:	Disposition): <u></u>	QA: N/C C	osed:		Date: _	
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DATE	STEP	Description of NC		Corrective Action Section		Verific	cation	Approval	Approval
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Item ID: Revision ID:	D412-698-04	41		*N900040			100)* s	setup Stai	rt *N	S1*
Item Name:	Hinge Panel A	Automatic Door Opener,	LH (Standard)						Sto	p *\	S2*
Start Date: Required Date: Reference:	07/06/2012 21/06/2012	Start Qty: 1.00 Req'd Qty: 1.00	*1* *1*		Cust Item I Customer:	D:				T W.	. 1/
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DATE	STEP	Description of NC Section A	Initial Chief Eng	Corrective Action Section B Action Description Chief Eng	Sign & Date	Verificati Section (Approval QC Inspector

NCR:		WORK ORDER NON-CONFORMANCE (NCR)										
		Description of NC		Corrective Action Section B	Verification	Approval	Approval					
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 Work Order ID:
 85382
 85382

 Parent Item:
 D412-698-041
 D412-698-041

 Parent Item Name:
 Hinge Panel Automatic Door Opener, LH (Standard)

Start Date: 07/06/2012

Required Date: 21/06/2012

Start Qty: 1.00

Required Qty: 1.00

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Comments:	IPP REV:A 11.08	3.18 new issue I	OD ver	f:JLM									
Component Item ID/ Item Name	Replacement Item ID	Mfg/ Purch		Primary Location	Last Location	Route Seq ID	Unit of Measure	Qty on Hand	Qty per Kit	Total Qty	Qty Issued	Date Issued	Status
D3552-21 *D3552-21 Door Prop	F1*	Manufactured	No		•	110	Each	3.0000	**	8525	21 fx	3	
-				Location		Loc (<u>Oty</u>	Loc Code					
				ST272			3		_		-		
					68168		1		_		-		
D3622-1	1	Manufactured	No		73948	110	2 Each	36.0000	1	ا ا	-		
*D3622-1	*								**		20		
				Location		Loc (<u>Oty</u>	Loc Code					¥
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					80405		1		_	70.0			
D4330-041		Manufactured	No		82072	110	35 Each	1.0000		8207	y		
> *D433020 Airframe Bracket Assemb)41* bly, LH								**		28		30
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*D4330-3 Door Bracket	3 *	Manufactured	No			110	Each	5.0000	**	1 0	JB		12/06
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Dart Aerospace Ltd

W/O:		-	WORK ORDER	CHANGES				
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Part No	:	PAR #:	Fault Category:	NCR: Ye	s No DQ	A:	Date: _	
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		Description of NC		Corrective Action Section B	Verification	Approval	Approval					
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Work Order ID: 85382

No

Parent Item: D412-698-041 *D412-698-041 *Parent Item Name: Hinge Panel Automatic Door Opener, LH (Standard)

Purchased

Purchased

Start Date: 07/06/2012

Required Date: 21/06/2012

Start Qty: 1.00

Required Qty: 1.00

MS21042L5

Location	Loc Qty	Loc Code
300	500	
121652	500	
ST300	889	
108827	8	
116105	5	
116548	43	
117611	18	
119109	807	
17651	8	
	110 Ea	ach 3,259.000

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No

Location Loc Qty ST298 3259 119546 1259 121255 2000

Loc Code

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NCR:		WORK ORDER NON-CONFORMANCE (NCR)										
		Description of NC	,	Corrective Action Section B	Verification	Approval	Approval					
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DART SERVICE INSTRUCTION TO AMEND INSTALLATION INSTRUCTIONS

IIN-D412-698 REV. E OR LATER APPROVED REVISION REF. TCCA STC: SH92-43

REF. FAA STC: SR01446NY REF. EASA STC: EASA.IM.R.S.00709 REF. BRASIL STC: 2010S10-09

1.0 PURPOSE:

The purpose of this DSI is to add the optional automatic door opener kit for both the OEM aluminum (Bell P/N 205-031-423-005/-157/-006/-158) and Dart Spacedoor™(Dart P/N D412-694-03/-04) hinge panel doors.

The Dart D412-698-XXX Hinge Panel Automatic Door Opener Kit permits the installation of a gas spring between the hinge panel doors (or "quarter" doors) and the cabin ceiling to prevent the doors from closing while the occupants enter

The D412-698-041 Hinge Panel Automatic Door Opener Kit installs on the LH side of the aircraft and the D412-698-042 Hinge Panel Automatic Door Opener Kit installs on the RH side of the aircraft. These kits are designed to mount onto the OEM aluminum door (Bell P/N 205-031-423-005/-157/-006/-158).

The D412-698-043 Hinge Panel Automatic Door Opener Kit installs on the LH side of the aircraft and the D412-698-044 Hinge Panel Automatic Door Opener Kit installs on the RH side of the aircraft. These kits are designed to mount onto the Dart Spacedoor™(Dart P/N D412-694-03/-04).

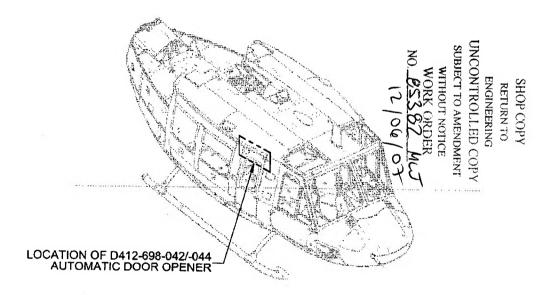


FIGURE 1: LOCATION OF HINGE PANEL AUTOMATIC DOOR OPENER KITS (RH Installation Shown, LH Opposite)

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DEPARTMENT OF TRANSPORT
AIRCRAFT CERTIFICATION BRANCH DAO # 01-O-01

APPRØVED D. SHEPHERD (DE # 02)

12.01,20 CERT. NO.: SH92-43 ISSUE NO .:

В	ADDED ITEMS FIGURE	NOTE TO TRIM 11/12 OF SECTI 3 ACCORDING	EXISTING TRIM/CEILING PANELS: ONS 3.0/4.0 (SHT 2/4); UPDATED SLY. REF: PAR11-150.	МВ	12.01.17	
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DATE 12.01.17			COPYRIGHT © 2011 BY DART THIS COCLIMENT IS PRIVATE AND CONFIDENTIAL AND IS SUPPLE NOT TO BE USED FOR ANY PURPOSE OR COPED OR COMMANS WHITTEN PERMISSION FROM IN OUR WHITTEN PERMISSION FRO	ED ON THE EXPRES CATED TO ANY DT	S CONDITION THAT IT IS	

3.0 INSTALLATION OF D412-698-041/-042 HINGE PANEL AUTOMATIC DOOR OPENER (OEM ALUMINUM DOOR):

NOTE: The following instructions are for a typical RH side installation. For installation on the LH side, substitute D4330-041 for D4330-042.

- Remove any ceiling or trim panels located on the RH foward cabin ceiling. Remove existing door spring Bell p/n 205-030-303-101 and associated hardware.
- Using Figures 2 and 3 and dimensions shown as reference, locate D4330-042 Airframe Bracket Assembly, RH
 on the intersection of the existing aircraft rivet lines. The angled part of D4330-042 Airframe Bracket
 Assembly, RH is required to pick up on a minimum of 3 rivets. The bracket may be hand formed as
 necessary to improve fit with the contoured ceiling.
- 3. Mark the location of D4330-3 Door Bracket on the Hinge Panel Door as shown in Figure 4.
- 4. Verify that the D3552-21 Gas Spring will close properly (0.25" travel before it is fully retracted) when the door is closed. Adjust the position of the D4330-042 Airframe Bracket Assembly, RH if required to maintain rivet edge distance.
- Verify that the D3552-21 Gas Spring will open the door to the desired position. Ensure the Gas Spring does not interfere with the door or airframe. Adjust location of D4330-3 Door Bracket if required.
 NOTE: When the door is fully open, it should be approximately perpendicular to the aircraft.
- 6. Remove the 205-031-423-005/-157/-006/-158 aluminum door from the aircraft per the Aircraft Maintenance Manual.
- Remove any existing rivets where the D4330-042 Airframe Bracket Assembly, RH and D4330-3 Door Bracket will be installed.
- Locate D4330-042 Airframe Bracket Assembly, RH on the aircraft ceiling per Figure 3. Transfer drill #30 (Ø 0.129")
 holes from the ceiling to D4330-042. Deburr holes.
- Locate D4330-3 Door Bracket on the hinge panel door per Figure 4. Using Figure 5 as a guide for the approximate locations of the mounting holes, transfer drill 3x #30 (Ø 0.129") holes from the door skin to D4330-3 and drill 2x additional holes at locations shown in Figure 5. Deburr holes.
- Apply a thin layer of Proseal 890 (or equivalent) to faying surface of the D4330-042/-3 brackets. Install the D4330-042 Airframe Bracket Assembly, RH using CR3213-4 rivets as shown in Figure 3. Install D4330-3 Door Bracket using CR3213-4 rivets as shown in Figure 4.
- 11. Pre-install the ceiling or trim panels that were removed in Step 1 and transfer mark contour of D4330-041 or -042 as shown in Figure 3 (SHT 5). Remove ceiling or trim panels and perform cutout to clear D4330-041 or -042. Touch-up finish of ceiling or trim panels cutout area in accordance with aircraft maintenance manual and re-install any ceiling or trim panels.

12.01.17

- 12. Install a D3622-1 Ball Stud on the D4330-3 Door Bracket using hardware specified in Figure 4.
- 13. Re-install modified door on aircraft per Aircraft Maintenance Manual.
- 14. Install D3552-21 Gas Spring onto the D3622-1 Ball Studs per Figure 2.
- 15. Verify proper gas spring and door operation.

CANADA
DEPARTMENT OF TRANSPORT
AIRCRAFT CERTIFICATION
BRANCH
DAO # 01-O-01

D. SHEPHERD (DE # 02)

DATE: CERT. NO.: ISSUE NO.: 12.01.20 SH92-43

DESIGN DART AEROSPACE LTD DRAWN HAWKESBURY, ONTARIO, CANADA DRAWING NO. CHECKED REV. B **DSI 9548** MFG. APPR. N/A SHEET 2 OF 8 APPROVED VA TITLE SCALE DE APPR. HINGE DOOR OPENER KIT NTS DATE

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4.0 INSTALLATION OF D412-698-043/-044 HINGE PANEL AUTOMATIC DOOR OPENER:

NOTE: The following instructions are for a typical RH side installation. For installation on the LH side, substitute D4330-041 for D4330-042.

- Remove any ceiling or trim panels located on the RH foward cabin ceiling. Remove existing door spring Dart p/n D3113-1 and associated hardware.
- Using Figures 2 and 3 and dimensions shown as reference, locate D4330-042 Airframe Bracket Assembly, RH
 on the intersection of the existing aircraft rivet lines. The angled part of D4330-042 Airframe Bracket
 Assembly, RH is required to pick up on a minimum of 3 rivets. The bracket may be hand formed as
 necessary to improve fit with the contoured ceiling.
- Mark the location of D4330-3 Door Bracket on the Spacedoor™ as shown in Figure 6.
- Verify that the D3552-21 Gas Spring will close properly (0.25" travel before it is fully retracted) when the door is closed. Adjust the position of the D4330-042 Airframe Bracket Assembly, RH if required to maintain rivet edge distance.
- Verify that the D3552-21 Gas Spring will open the door to the desired position. Ensure the Gas Spring
 does not interfere with the door or airframe. Adjust location of D4330-3 Door Bracket if required.
 NOTE: When the door is fully open, it should be approximately perpendicular to the aircraft.
- 6. Remove the D412-694-03/-04 Door Assembly from the aircraft per ICA-D412-694 Section 52.1.
- 7. Remove any existing rivets where the D4330-042 Airframe Bracket Assembly, RH will be installed.
- 8. Locate D4330-042 Airframe Bracket Assembly, RH on the aircraft ceiling per Figure 3. Transfer drill #30 (ϕ 0.129") holes from the ceiling to D4330-042. Deburr holes.
- Locate D4330-3 Door Bracket on the Spacedoor™ per Figure 6. Using Figure 7 as a guide for the approximate locations of the mounting holes, drill 3x #7 (Ø0.201") holes in D4330-3 and the inner door skin. Enlarge holes in Spacedoor™ inner skin to 9/16" (Ø0.5625").
- Install the 3x 80-005-2-8 inserts per Paragraph 3.2.12 of the BHT-MED-SRM at the locations drilled in Step 9 in in the Spacedoor™ using Hysol EA934NA.
- Apply a thin layer of Proseal 890 (or equivalent) to faying surface of the D4330-042/-3 brackets. Install the D4330-042 Airframe Bracket Assembly, RH using CR3213-4 rivets as shown in Figure 3. Install D4330-3 Door Bracket using hardware as shown in Figure 6.
- Pre-install the ceiling or trim panels that were removed in Step 1 and transfer mark contour of D4330-041 or -042 as shown in Figure 3 (SHT 5). Remove ceiling or trim panels and perform cutout to clear D4330-041 or -042. Touch-up finish of ceiling or trim panels cutout area in accordance with aircraft maintenance manual and re-install any ceiling or trim panels.
- 13. Install a D3622-1 Ball Stud on the D4330-3 Door Bracket using hardware specified in Figure 6.
- 14. Re-install modified door on aircraft per ICA-D412-694 Section 52.2.
- 15. Install D3552-21 Gas Spring onto the D3622-1 Ball Studs per Figure 2.
- 16. Verify proper gas spring and door operation.

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AIRCRAFT CERTIFICATION
BRANCH
DAO #01-0-01

APPROVED

BY:
D. SHEPHERD (DE #02)

DATE:
CERT. NO.:
SH92-43
ISSUE NO.:
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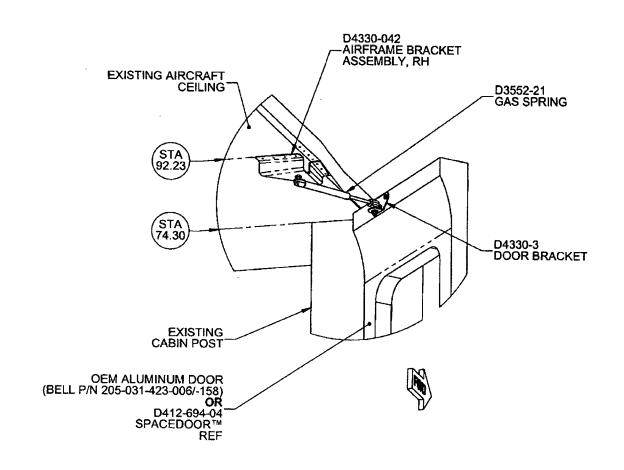


FIGURE 2: INSTALLATION DETAIL (RH Installation Shown, LH Opposite)

12.01.17

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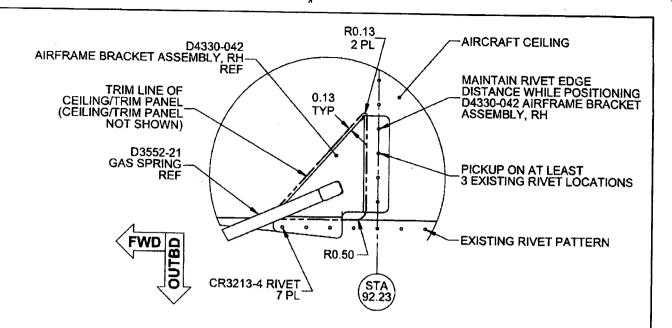
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FIGURE 3: LOCATION OF D4330-042 AIRFRAME BRACKET ASSEMBLY (RH Shown, LH Opposite)

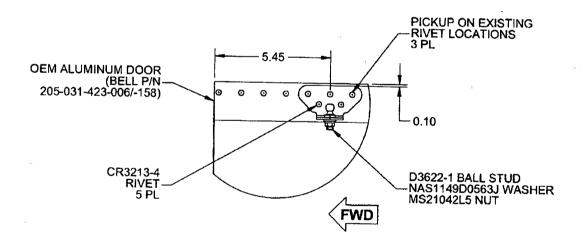


FIGURE 4: LOCATION OF D4330-3 DOOR BRACKET (OEM ALUMINUM DOOR) (RH Shown, LH Opposite)

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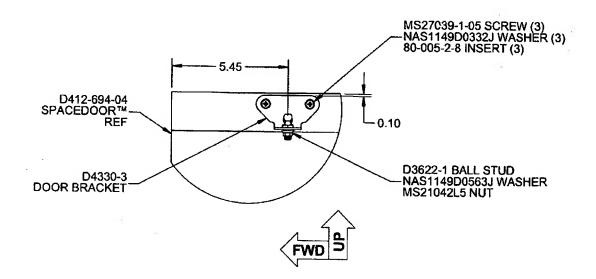


FIGURE 6: LOCATION OF D4330-3 DOOR BRACKET (SPACEDOOR TM) (RH Shown, LH Opposite)

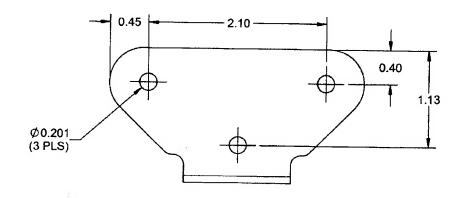


FIGURE 7: LOCATION OF D4330-3 DOOR BRACKET MOUNTING HOLES (SPACEDOOR™)

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DAO # 01-0-01

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CATED TO ANY OTHER PERSON WITHOUT

5.0 WEIGHT & BALANCE:

INSTALLATION	WEIGHT	LATERAL		LONGITUDINAL		
		Arm	Moment	Arm	Moment	
D412-698-041	0.23 lb	-43.63 in	-10.03 in-lb	86.27 in	19.84 in-lb	
LH (Standard)	0.10 kg	-1.11 m	-0.11 m-kg	2.19 m	0.22 m-kg	
D412-698-042	0.23 lb	43.63 in	10.03 in-lb	86.27 in	19.84 in-lb	
RH (Standard)	0.10 kg	1.11 m	0.11 m-kg	2.19 m	0.22 m-kg	
D412-698-043	0.23 lb	-43.63 in	-10.03 in-lb	86.27 in	19.84 in-lb	
LH (Spacedoor™)	0.10 kg	-1.11 m	-0.11 m-kg	2.19 m	0.22 m-kg	
D412-698-044	0.23 lb	43.63 in	10.03 in-lb	86.27 in	19.84 in-lb	
RH (Spacedoor™)	0.10 kg	1.11 m	0.11 m-kg	2.19 m	0.22 m-kg	

6.0 PARTS LIST:

Qty -041	Qty -042	Qty -043	Qty -044	Part Number	Description
X				D412-698-041	HINGE PANEL AUTOMATIC DOOR OPENER KIT, LH (STANDARD)
	Х			D412-698-042	HINGE PANEL AUTOMATIC DOOR OPENER KIT, RH (STANDARD)
		X		D412-698-043	HINGE PANEL AUTOMATIC DOOR OPENER KIT, LH (SPACEDOOR™)
			X	D412-698-044	HINGE PANEL AUTOMATIC DOOR OPENER KIT, RH (SPACEDOOR™)
1	1	1	1	D3552-21	GAS SPRING
1	1	_1	1	D3622-1	BALL STUD
1		1		D4330-041	AIRFRAME BRACKET ASSEMBLY, LH
	1]	1	D4330-042	AIRFRAME BRACKET ASSEMBLY, RH
1	1	1	_1	D4330-3	DOOR BRACKET
		3	3	80-005-2-8	INSERT
1	1	1		MS21042L5	NUT
		3	3	MS27039-1-05	SCREW
		3			WASHER
1	1	1	1		WASHER

CANADA
DEPARTMENT OF TRANSPORT
AIRCRAFT CERTIFICATION
BRANCH DAO # 01-O-01

APPROVED BY: D. SHEPHERD (DE # 02)

DATE: CERT. NO.: ISSUE NO.:

12.01.20 SH92-43

DESIGN	DE
DRAWN	5
CHECKED	(MA)
MFG. APPR.	N/A
APPROVED	M
DE APPR.	
DATE 12.0	1.17

	DART AEROSPA	
	HAWKESBURY, ONTAR	IO. CANADA
ı	DRAWING NO.	
		REV. B
	DSI 9548	
		CHEET 9 OF 9

SHEET 8 OF 8 TITLE

HINGE DOOR OPENER KIT

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SCALE

5.0 WEIGHT & BALANCE:

INSTALLATION	WEIGHT	LAT	ERAL	LONGI	FUDINAL
INSTALLATION	***	Arm	Moment	Arm	Moment
D412-698-041	0.23 lb	-43.63 in	-10.03 in-lb	86.27 in	19.84 in-lb
LH (Standard)	0.10 kg	-1.11 m	-0.11 m-kg	2.19 m	0.22 m-kg
D412-698-042 · RH (Standard)	0.23 lb	43.63 in	10.03 in-lb	86.27 in	19.84 in-lb
	0.10 kg	1.11 m	0.11 m-kg	2.19 m	0.22 m-kg
D412-698-043	0.23 lb	-43.63 in	-10.03 in-lb	86.27 in	19.84 in-lb
LH (Spacedoor™)	0.10 kg	-1.11 m	-0.11 m-kg	2.19 m	0.22 m-kg
D412-698-044	0.23 lb	43.63 in	10.03 in-lb	86.27 in	19.84 in-lb
RH (Spacedoor™)	0.10 kg	1.11 m	0.11 m-kg	2.19 m	0.22 m-kg

6.0 PARTS LIST:

Qty -041	Qty -042	Qty -043	Qty -044	Part Number	Description
·X				D412-698-041	HINGE PANEL AUTOMATIC DOOR OPENER KIT, LH (STANDARD)
	X			D412-698-042	HINGE PANEL AUTOMATIC DOOR OPENER KIT, RH (STANDARD)
		Х		D412-698-043	HINGE PANEL AUTOMATIC DOOR OPENER KIT, LH (SPACEDOOR™)
			X	D412-698-044	HINGE PANEL AUTOMATIC DOOR OPENER KIT, RH (SPACEDOOR™)
11	1	1	1	D3552-21	GAS SPRING
7 1	1	1	1	D3622-1	BALL STUD
/1		1		D4330-041	AIRFRAME BRACKET ASSEMBLY, LH
	1		1	D4330-042	AIRFRAME BRACKET ASSEMBLY, RH
/1	1	1	1	D4330-3 /	DOOR BRACKET
		3_	3	80-005-2-8	INSERT
<u> </u>	_1_	-1-	_1_	MS21042L5	NUT
		3	3	MS27039-1-05	SCREW
		3	_3	NAS1149D0332J	WASHER
5	_1_	_1_	-1-	NAS1149D0563J	WASHER

CANADA
DEPARTMENT OF TRANSPORT
AIRCRAFT CERTIFICATION
BRANCH
DAO # 01-O-01

BY: D. SHEPHERD (DE #02)

DATE: 12.01.20
CERT. NO.: SH92-43
ISSUE NO.: 5

DESIGN	DE	DART AEROSPACE LTD	
DRAWN	1	HAWKESBURY, ONTARIO, CANADA	
CHECKED	(NA)	DRAWING NO.	REV. B
MFG. APPR.	N/A	DSI 9548 SHEET	8 OF 8
APPROVED	Ma	TITLE	SCALE
DE APPR.	4	HINGE DOOR OPENER KIT	NTS
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